

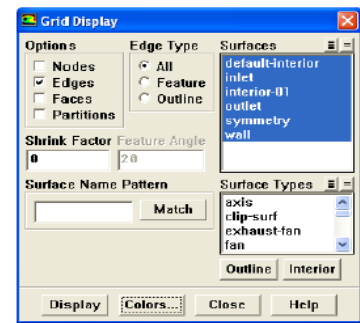
Step by Step Guide for Boundary Layer on a Flat Plate

I. Preparation

1. Download the file “**Flow-Over-FlatPlate.msh**” from Kalam and save it to your working folder
2. Open workbench and drag “**Fluid Flow (Fluent)**” toolbox to the “**Project Schematic**” window
3. Right click on **Mesh > Import mesh file > Browse** and select the saved mesh file.
4. Click the Setup to start the 2D double precision solver of FLUENT.

II. General

1. Display the grid (See the Figure). Make sure all the surfaces are highlighted.
Display > mesh...
2. Click **Close** and close the Grid Display panel.
3. Click the Report Quality panel. Make sure that the orthogonal quality is close to 1.



III. Modeling

1. Since problem to be solved is laminar, retain the default solver settings.
 Click **Models** on the left menu then **Viscous > Laminar...**

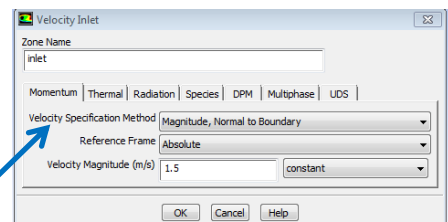
IV. Defining Materials

1. retain the default properties of air
Materials...

V. Boundary Conditions

Click **Boundary Conditions** on the left menu

1. Set the boundary condition for velocity inlet (**inlet**).
 - a. Select **inlet** from the Zone selection list.
 - b. Select **Velocity-inlet** from the Type selection list
 - c. Click the **Edit...** button to open the Velocity-inlet panel.



- d. Enter 1.5 m/s for **Velocity Magnitude**.
 - e. Click **OK** to close the Velocity Inlet panel.
2. Retain the default boundary conditions for pressure outlet (**outlet**).
 - a. Select **outlet** from the Zone selection list.
 - b. Select **pressure-outlet** from the Type selection list.
 - c. Click the **Edit...** button to open the Pressure Outlet panel.
 - d. Make sure that the **Gauge Pressure** value is 0.
 3. Retain the default boundary conditions for the plate (**wall**).
 4. Set the boundary condition for top.
 - a. Select **top** from the Zone selection list.
 - b. Select **symmetry** from the Type selection list.

VI. Reference Values

Click **Reference Values** on the left menu

Select **inlet** under **Compute from**

VII. Solution

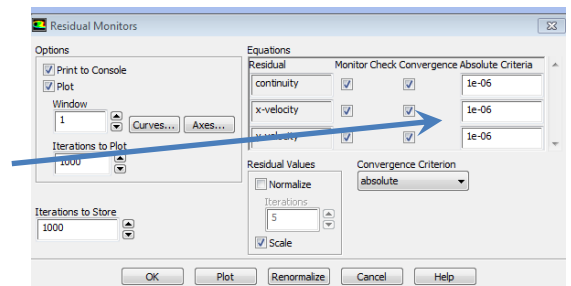
1. Set the solution methods.

Click on **Method** on the left menu

- a. Select **second order upwind** for momentum from the **Spatial Discretization** drop-down lists.
 - b. Retain the remaining
2. Set the convergence criteria.

Click **Monitors** on the left menu

- a. Double Click **Residuals-print, plot** from the “**Residuals, statistics and force monitors**” drop-down list.
- b. Enable **Plot** in the **Options** group box.
- c. Enter **1e-06** for Absolute Criteria for all the equations.



Higher order discretization schemes and tighter convergence criteria are desirable for accurate resolution of boundary thermal boundary layer.

- d. Click **OK** to close the **Residual Monitors** panel.
3. Initialize the flow.
Click **Initialization** on the left menu
 - a. Select **Standard Initialization** from the **Initialization Methods** list
 - b. Select **inlet** from the “**Compute From**” drop-down list.
It will update values of all the variables based on the boundary conditions at the inlet. Make sure that all values are correct.
 - c. Click **Initialize**.
For faster convergence, set all the velocities to zero.
 4. Save the case file (filename.cas.gz).
File > Write > Case...
Retain the default Write Binary Files option so that you can write a binary file.
 5. Start the calculation by requesting 1000 iterations.
Solve > Run Calculation...
 - a. Set Number of Iterations to 1000.
 - b. Click **Calculate**.
 - c. Close the **Calculate** panel

VIII. Postprocessing

1. Display pressure and velocity contours and vectors.
Display > Graphics and Animations...
 - a. Double click **Contours** from the **Graphics and Animation** drop-down lists
 - b. Select **Pressure...** and **Static Pressure** from the “**Contours of**” drop-down lists.
 - c. Enable **Filled** from the **Options** list.
 - d. Click **Display**.

Zoom in to get the complete temperature distribution of temperature along the length of plate

Similarly, for the velocity contour:

- e. Select **Velocity...** and **Velocity magnitude** from the “**Contours of**” drop-down lists.
- f. Enable **Filled** from the **Options** list.
- g. Click **Display**.

Zoom in to get the complete temperature distribution of temperature along the length of plate

2. Display the velocity vectors.

- a. Double click **Vectors** from the **Graphics and Animation** drop-down lists
- b. Select **Velocity...** and **Velocity Magnitude** from the “**Vector of**” drop-down lists.
- c. Enable **Filled** from the **Options** list.
- d. Click **Display**.

Zoom in to get the complete temperature distribution of temperature along the length of plate

3. Generate lines at $x = 0.1$ m , $x = 0.2$ m, $x = 0.3$ m and at $x = 0.4$ m ,.... Up to $x = 1$ m

- a. You need to examine the boundary layer profiles and boundary layer thicknesses in detail at all x locations along the flat plate. As an example five locations namely at inlet ($x=0$, $x = 0.2$ m, 0.4 m, $x = 0.6$ m, 0.8 m, and outlet ($x=1$ m) from the leading edge are explained below.
- b. In the main FLUENT menu, click **Surface-Line/Rake**. Type in the desired starting and ending x and y locations of the vertical line, i.e. a vertical line going from $(0.2, 0)$ to $(0.2, 0.5)$.
- c. The New Surface Name should be assigned at this point. It is suggested that this line be called “ **$x=0.2$ -m-profile**” or something descriptive of its intended purpose.
- d. Click **Create** to create the line.
- e. Similarly, create a lines at $x = 0.4$, $x = 0.6$ and $x = 0.8$; suggested label are “ **$x=0.4$ -m-profile**”, “ **$x=0.6$ -m-profile**”, and “ **$x=0.8$ -m-profile**”.
- f. **Close** Line/Rake Surface.
- g. To view these newly created lines, click **Display>Mesh**. Under “**Surfaces**”, Left Mouse Click (LMC) the default interior called “**interior-surface_body**” to unselect it. LMC the names of the newly created lines to select them instead.

- h. Click **Display**. The lines should be visible at the appropriate locations. If not, create them again more carefully.
 - i. Close the Mesh Display window.
4. Plot the velocity profiles at inlet ($x=0$), $x = 0.2$ m , $x = 0.6$ m, at $x = 0.1$ m , and at outlet ($x=1$ m)

Display > Plots...

- a. Double click **XY Plots** from the **Plots** drop-down lists
- b. Select (highlight) the lines from the list of **surfaces** on the right side under Results, Plots-XY Plot-Set Up.
- c. select **Velocity** from the **X Axis Function**,
- d. Also select **outlet**, which is at $x = 1$ m.
- e. Click **Plot**
- f. In the upper left corner of the window, turn off (uncheck) **Position on X Axis**, and turn on (check) **Position on Y Axis**. This will make the vertical axis the y position on the plot, as desired.
- g. In the upper middle part of that window, set **Plot Direction** to $X = 0$ and $Y = 1$. This will make the Y- coordinate position appear on the vertical axis, as desired for a standard velocity profile plot.
- h. The boundary layer profiles should be there, but may be difficult to see since the vertical axis extends all the way to the upper boundary of the computational domain. The axes limits can be changed as follows:
 - i. Click **Axes...**
 - j. Choose **Y**.
 - k. Unselect **Auto Range**, and select **Major Rules**. Set Range from “0” to “0.06” m.
 - l. **Apply** then **close**.

Reference

FLUENT 6.1 Tutorial Guide, February 2003,
file:///C:/Users/FKM%2311/Downloads/documents.mx_fluent-tutorials-55844f4e977b8.pdf