



BFF1113 Engineering Materials



DR. NOOR MAZNI ISMAIL FACULTY OF MANUFACTURING ENGINEERING



Course Guidelines:

- 1. Introduction to Engineering Materials
- 2. Bonding and Properties
- 3. Crystal Structures & Properties
- 4. Imperfection in Solids
- 5. Mechanical Properties of Materials
- 6. Physical Properties of Materials
- 7. Failure & Fundamental of Fracture
- 8. Metal Alloys
- 9. Phase Diagram
- 10. Phase Transformation Heat Treatment
- 11. Processing and Application of Metals
- 12. Ceramic Materials
- 13. Polymer Materials
- 14. Composite Materials
- 15. Corrosion & Degradation of Materials
- 16. Environment and Sustainability





Failure & Fundamental of Fracture

- 1. Fundamental of Fracture
- 2. Principle of fracture mechanics
- 3. Factors of Fracture





Fundamental of Fracture

Ductile fracture Brittle fracture





Fundamental of Fracture

Fracture:

- Separation of body into 2 or more piece in response to stress.

- Influences the selection of a material for a application, the methods of manufacturing, and the service life of the component.

Two modes of fracture:

Ductile fracture:

- Occurs with plastic deformation
- High energy absorption

Brittle fracture

- Little or no plastic deformation
- Catastrophic
- Low energy absorption



Schematic illustration of types of failures in materials. (a) necking and fracture of ductile materials. (b) buckling of ductile material under compressive load. (c) fracture of brittle material in compression. (d) cracking on the barrel surface of ductile materials in compression.





Ductile vs Brittle Failure



UMP OPEN COURSEWARE



Ductile vs. Brittle Failure



cup-and-cone fracture

brittle fracture



Moderately Ductile Failure

 2 step in fracture: crack formation and crack propagation Evolution to failure:



UMP OPEN COURSEWARE

fibrous appearance

Example: Failure of a Pipe



 Ductile failure: one piece large deformation

 Brittle failure: many pieces small deformation





Intergranular (between grains)



304 S. Steel (metal)

Reprinted w/permission from "Metals Handbook", 9th ed, Fig. 633, p. 650. Copyright 1985, ASM International, Materials Park, OH. (Micrograph by J.R. Keiser and A.R. Olsen, Oak Ridge National Lab.)

Transgranular (through grains) 316 S. Steel (metal)

Reprinted w/permission from "Metals Handbook", 9th ed, Fig. 650, p. 357. Copyright 1985, ASM International, Materials Park, OH. (Micrograph by D.R. Diercks, Argonne National Lab.)





Polypropylene (polymer)

Reprinted w/permission from R.W. Hertzberg, "Defor-mation and Fracture Mechanics of Engineering Materials", (4th ed.) Fig. 7.35(d), p. 303, John Wiley and Sons, Inc., 1996.

Al Oxide (ceramic)

Reprinted w/ permission from "Failure Analysis of Brittle Materials", p. 78. Copyright 1990, The American Ceramic Society, Westerville, OH. (Micrograph by R.M. Gruver and H. Kirchner.)



Chapter 8 -

(Orig. source: K. Friedrick, *Fracture* 1977, Vol. 3, ICF4, Waterloo, CA, 1977, p. 1119.)

UMP OPEN COURSEWARE

Intragranular-crack propagation through the interior of grain

SEM Micrograph (transgranular) Path of crack propagation Grains (a)

COURSEWARE

Intergranular-crack propagation along the grain boundaries







Principle of fracture mechanics

- Stress concentration
 - Crack propagation
- Fracture toughness





STRESS CONCENTRATION POINT



Max stress occur at the crack tip Can be calculated by

$$\sigma_m = 2\sigma_o \left(\frac{a}{\rho_t}\right)^{1/2} = K_t \sigma_o$$

where $\rho_t = \text{radius of curvature}$ $\sigma_o = \text{applied stress}$ $\sigma_m = \text{stress at crack tip}$

• Stress conc. factor: $K_t = \sigma_{max} / \sigma_o$

The ratio of max stress over applied stress

Stress concentration cont.





Avoid sharp corners!





Crack propagation



WHEN DOES A CRACK PROPAGATE?





• Large K_t promotes failure: NOT \uparrow \uparrow SO \bigcirc K_t=3 BAD! \frown K_t>>3 \downarrow \downarrow





Crack propagation cont.

Criterion for Crack Propagation

Crack propagates if crack-tip stress (σ_m) exceeds a critical stress (σ_c)

i.e.,
$$\sigma_m > \sigma_c$$
 $\sigma_c = \left(\frac{2E\gamma_s}{\pi \alpha}\right)^{1/2}$

where

- E = modulus of elasticity
- γ_s = specific surface energy
- a = one half length of internal crack

For ductile materials => replace γ_s with $\gamma_s + \gamma_p$ where γ_p is plastic deformation energy



EXAMPLE PROBLEM M.1

Maximum Flaw Length Computation

A relatively large plate of a glass is subjected to a tensile stress of 40 MPa. If the specific surface energy and modulus of elasticity for this glass are 0.3 J/m^2 and 69 GPa, respectively, determine the maximum length of a surface flaw that is possible without fracture.

Solution

To solve this problem it is necessary to employ Equation M.3. Rearrangement of this expression such that *a* is the dependent variable, and realizing that $\sigma = 40$ MPa, $\gamma_s = 0.3 \text{ J/m}^2$, and E = 69 GPa, lead to

$$a = \frac{2E\gamma_s}{\pi\sigma^2}$$

= $\frac{(2)(69 \times 10^9 \text{ N/m}^2)(0.3 \text{ N/m})}{\pi (40 \times 10^6 \text{ N/m}^2)^2}$
= $8.2 \times 10^{-6} \text{ m} = 0.0082 \text{ mm} = 8.2 \ \mu\text{m}$



FRACTURE TOUGHNESS



a property that is a measure of a material's resistance to brittle fracture when a crack is present.

Design Against Crack Growth

· Crack growth condition:

 $K \ge K_c = Y_{\sigma} \sqrt{\pi a}$

• Y is a dimensionless parameter / function that depends on both crack and specimen sizes, and geometries as well as the manner of load application.

--Scenario 1: Max. flaw size dictates design stress.



--Scenario 2: Design stress dictates max. flaw size.



GEOMETRY, LOAD, & MATERIAL

Condition for crack propagation:

Stress Intensity Factor: Depends on load & geometry. Fracture Toughness: Depends on the material, temperature, environment, & rate of loading.

COURSEWAR

Values of K for some standard loads & geometries:



 $K \ge K_c$

EXAMPLE:



Design Example: Aircraft Wing

- Material has K_c = 26 MPa-m^{0.5}
- Two designs to consider...



Design A largest flaw is 9 mm failure stress = 112 MPa

Design B use same material largest flaw is 4 mm failure stress = ?

• Key point: Y and K_c are the same in both designs.

Result:

Use...

$$\begin{array}{c} 112 \text{ MPa } 9 \text{ mm} \\ (\sigma_c \sqrt{a_{\text{max}}}) \\ A = (\sigma_c \sqrt{a_{\text{max}}}) \\ Answer: (\sigma_c)_{B} = 168 \text{ MPa} \\ Reducing flaw size pays off! \end{array}$$



Factors of Fracture

Effects of inclusion

Transition temperature





1.0: Effects of Inclusions

- Inclusion of impurities have an influence on ductile fracture and the workability of materials
- Consist of impurities of various kinds, ex: particles
- Voids and porosity can develop during processing of metals
- 2 factors affect void formation:
 - 1. Bond between an inclusion and the matrix
 - 2. Hardness of the inclusion



2.0: Transition Temperature Influence of Temperature on Impact Energy

• Ductile-to-Brittle Transition Temperature (DBTT)...





Universiti Malavsia

Design Strategy: Stay Above The DBTT!

• Pre-WWII: The Titanic



Reprinted w/permission from R.W. Hertzberg, "Deformation and Fracture Mechanics of Engineering Materials", (4th ed.) Fig. 7.1(a), p. 262, John Wiley and Sons, Inc., 1996. (Orig. source: Dr. Robert D. Ballard, *The Discovery of the Titanic.*) • WWII: Liberty ships



Reprinted w/permission from R.W. Hertzberg, "Deformation and Fracture Mechanics of Engineering Materials", (4th ed.) Fig. 7.1(b), p. 262, John Wiley and Sons, Inc., 1996. (Orig. source: Earl R. Parker, "Behavior of Engineering Structures", Nat. Acad. Sci., Nat. Res. Council, John Wiley and Sons, Inc., NY, 1957.)

 Problem: Steels were used having DBTT's just below room temperature.

