## THEORY OF STRUCTURES CHAPTER 2 : DEFLECTION (MACAULAY METHOD) PART 1

by<br>Saffuan Wan Ahmad<br>Faculty of Civil Engineering \& Earth Resources saffuan@ump.edu.my



## Chapter 2 : Part 1 - Macaulay Method

- Aims
- Draw elastic curve for beam
- Write equation for bending moment
- Determine the deflection of statically determinate beam by using Double Integration Method.
- Write a single equation for bending moment.
- Determine the deflection of statically determinate beam by using Macaulay's Method.
- Expected Outcomes:
- Able to analyze determinate beam - deflection and slope by Macaulay Method.
- References
- Mechanics of Materials, R.C. Hibbeler, 7th Edition, Prentice Hall
- Structural Analysis, Hibbeler, 7th Edition, Prentice Hall
- Structural Analysis, SI Edition by Aslam Kassimali,Cengage Learning
- Structural Analysis, Coates, Coatie and Kong
- Structural Analysis - A Classical and Matrix Approach, Jack C. McCormac and James K. Nelson, Jr., 4th Edition, John Wiley


## WHAT IS DEFLECTION????



## INTRODUCTION

- deflection is a term that is used to describe the degree to which a structural element is displaced under a load.




## THE ELAS I IC CURVE

-The deflection diagram of the longitudinal axis that passes through the centrfold of each cross-sectional area of the beam

- Support that resist a force, such as pinned, restrict displacement
- Support that resist a moment such as fixed, resist rotation or slope as well as displacement.


## Example



Figure 1


Figure 2

## Example



A
Figure 1


Figure 2

## Example



Figure 1


Figure 2

## Example



Figure 1


$$
\Delta_{B}=0
$$

Figure 2

## Example



Figure 1


Figure 2

## Example


$\theta_{B} \neq 0$
$\theta_{C} \neq 0$
$\Delta_{A}=0$
$\Delta_{B}=0$
$\Delta_{C} \neq 0$
Figure 2

Three basic methods to find deflection for staticall Mineand Malaysia
PAHANG determinate beams:

by Saffuan Wan Ahmad
EULER - BERNNO ULLI THEEORY

- Also known as elastic-beam theory
- This theory form important differential equation that relate the internal moment in a beam to the displacement and slope of its elastic curve.
- This equation form the basis for the deflection methods.

$$
\frac{d^{2} v}{d x^{2}}=\frac{M}{E I}
$$

Equation 1

## THE DOUBLE INTERGRATION METHOD

- Moment, M is known expressible as a function of position $x$, the successive integrations of Eq . 1 will yield the beam's slope, $\theta$.

$$
\theta=\frac{d v}{d x}=\int \frac{M}{E I} d x
$$

- And the equation of the elastic curve, $v$ (displacement)

$$
v=f(x)=\iint \frac{M}{E I} d x
$$

## THE DOUBLE INTERGRATION METHOD

- This method depend on the loading of the beam.
- All function for moment must be written each valid within the region between discontinuities.
- Using equation 1 and the function for M , will give the slope and deflection for each region of the beam for which they are valid.


## THE DOUBLE INTERGRATION METHOD

- EXAM PLE

Consider a simply supported beam AB of length L and carrying concentrated load P at mid span, C as shown below. Use the double integration method. Find the equation of the elastic curve. El is constant.


- Function for the beam moment
- For span $0<x_{1} \varangle / 2$

$$
\mathrm{M}_{1}=\frac{\mathrm{P}}{2} x_{1}
$$

- For span $\mathrm{L} / 2<x_{2} \llbracket$

$$
\begin{gathered}
\mathrm{M}_{2}=\frac{\mathrm{P}}{2} x_{2}-\mathrm{P}\left(x_{2}-\frac{\mathrm{L}}{2}\right) \\
\mathrm{M}_{2}=-\frac{\mathrm{P}}{2} x_{2}+\frac{\mathrm{PL}}{2}
\end{gathered}
$$

- Replace $M_{1}$ into slope and displacement integration.

$$
\begin{gathered}
\frac{d^{2} v}{d x^{2}}=\frac{\frac{\mathrm{P}}{2} x_{1}}{E I} \\
\theta_{1} E I=\frac{d v}{d x}=\int \frac{\mathrm{P}}{2} x_{1} d x \\
v_{1} E I=f(x)=\iint \frac{\mathrm{P}}{2} x_{1} d x
\end{gathered}
$$

$$
\begin{aligned}
& \theta_{1} E I=\int \frac{\mathrm{P}}{2} x_{1} d x \\
& \theta_{1} E I=\frac{\mathrm{P} x_{1}^{2}}{4}+\mathrm{C}_{1}
\end{aligned}
$$

Then,

$$
\begin{aligned}
& v_{1} E I=\int \frac{\mathrm{P} x_{1}^{2}}{4}+\mathrm{C}_{1} d x \\
& v_{1} E I=\frac{\mathrm{P} x_{1}^{3}}{12}+\mathrm{C}_{1} x_{1}+\mathrm{C}_{2}
\end{aligned}
$$

- Here we have 2 unknown $\mathrm{C}_{1}$ and $\mathrm{C}_{2}$
- Replace $M_{2}$ into slope and displacement integration.

$$
\begin{gathered}
\frac{d^{2} v}{d x^{2}}=\frac{-\frac{\mathrm{P}}{2} x_{2}+\frac{\mathrm{PL}}{2}}{E I} \\
\theta_{2} E I=\frac{d v}{d x}=\int-\frac{\mathrm{P}}{2} x_{2}+\frac{\mathrm{PL}}{2} d x \\
v_{2} E I=f(x)=\iint-\frac{\mathrm{P}}{2} x_{2}+\frac{\mathrm{PL}}{2} d x
\end{gathered}
$$

$$
\begin{aligned}
\theta_{2} E I & =\int-\frac{\mathrm{P}}{2} x_{2}+\frac{\mathrm{PL}}{2} d x \\
\theta_{2} E I & =-\frac{\mathrm{P}}{4} x_{2}^{2}+\frac{\mathrm{PL}}{2} x_{2}+\mathrm{C}_{3}
\end{aligned}
$$

Then,

$$
\begin{gathered}
v_{2} E I=\int-\frac{\mathrm{P}}{4} x_{2}^{2}+\frac{\mathrm{PL}}{2} x_{2}+\mathrm{C}_{3} d x \\
v_{2} E I=-\frac{\mathrm{P}}{12} x_{2}^{3}+\frac{\mathrm{PL}}{4} x_{2}^{2}+\mathrm{C}_{3} x_{2}+\mathrm{C}_{4}
\end{gathered}
$$

- Here we have 2 unknown $C_{3}$ and $C_{4}$
- Using boundary conditions

$$
\begin{gathered}
-v_{1}=0, x_{1}=0 \\
-v_{2}=0, x_{2}=\mathrm{L} \\
-x_{1}=x_{2}=\frac{\mathrm{L}}{2} \\
\cdot v_{1}=v_{2} \\
\cdot \theta_{1}=\theta_{2}
\end{gathered}
$$

- This will solve $\mathrm{C}_{1}, \mathrm{C}_{2}, \mathrm{C}_{3}$ and $\mathrm{C}_{4}$

Solving all the unknown, $\mathrm{C}_{\mathrm{x}}$, will give slope and displacement for the element.
-At the support $x_{1}=0, x_{2}=\mathrm{L}$

$$
\theta_{1}=\theta_{2}= \pm \frac{P L^{2}}{16}
$$

-At the mid $\operatorname{span} x_{1}=x_{2}=\frac{\mathrm{L}}{2}$

$$
v_{1}=v_{2}=-\frac{P L^{3}}{48}
$$

## MACAULAY'S METHOD

Mac-Caulay's method is a means to find the equation that describes the deflected shape of a beam
From this equation, any deflection of interest can be found

- Mac-Caulay's method enables us to write a single equation for bending moment for the full length of the beam
When coupled with the Euler-Bernoulli theory, we can then integrate the expression for bending moment to find the equation for deflection using the double integration method.
M acauly's Method allow us to 'turn off' partial of moment function when the value inside a bracket in that function is zero or negative.
(CC) (i) (8) (O) NO SA


## Macaulay's Method

In this method, the moment function only will be considered at end of the section


$$
P(x-a)^{1}
$$



$$
\frac{w(x-a)^{2}}{2}
$$



$$
M_{0}(x-a)^{0}
$$

## M acauly's M ethod

Let us again consider a simply supported beam AB of length L and carrying concentrated load $P$ at mid span, C as shown below. El is constant. This example are going to show how to find the equation of elastic curve for the beam by 'turn off' part of a function using $M$ acauly's $M$ ethod.


- Again we must write a function for the beam moment that can describe the moment for the beam wholly from the left side.
- This beam have 2 span. M acauly's M ethod will use the moment function to the very right with only $x$ function as distance. Where here for example:
- Span L/2<xব

$$
\mathrm{M}=\frac{\mathrm{P}}{2} x-\mathrm{P}\left(x-\frac{\mathrm{L}}{2}\right)
$$

- Take note here M acauly's M ethod use a different bracket that have a special function that have an advanced understanding and application.
- Span L/2<x 4

$$
\mathrm{M}=\frac{\mathrm{P}}{2} x-\mathrm{P}\left(x-\frac{\mathrm{L}}{2}\right)
$$

- The bracket above allow the function of 'turn off' when inside value is negative or zero.
- Means if we have $x \leq \frac{\mathrm{L}}{2}$ the $\left\langle x-\frac{\mathrm{L}}{2}\right\rangle$ will be zero
- Mathematically explained as :

$$
\left\langle x-\frac{\mathrm{L}}{2}\right\rangle=\left\{\begin{array}{cc}
0, & x \leq \frac{\mathrm{L}}{2} \\
x-\frac{\mathrm{L}}{2}, & x>\frac{\mathrm{L}}{2}
\end{array}\right.
$$

- Applying Euler-Bernoulli Theory replace M into slope and displacement integration.

$$
\begin{gathered}
E I \frac{d^{2} v}{d x^{2}}=\frac{\mathrm{P}}{2} x-\mathrm{P}\left(x-\frac{\mathrm{L}}{2}\right) \\
\theta E I=\frac{d v}{d x}=\int \frac{\mathrm{P}}{2} x-\mathrm{P}\left(x-\frac{\mathrm{L}}{2}\right) d x \\
v E I=f(x)=\iint \frac{\mathrm{P}}{2} x-\mathrm{P}\left\langle x-\frac{\mathrm{L}}{2}\right\rangle d x
\end{gathered}
$$

- From the slope integration:

$$
\begin{aligned}
& \theta E I=\int \frac{\mathrm{P}}{2} x-\mathrm{P}\left\langle x-\frac{\mathrm{L}}{2}\right\rangle d x \\
& \theta E I=\frac{\mathrm{P}}{4} x^{2}-\frac{\mathrm{P}}{2}\left(x-\frac{\mathrm{L}}{2}\right\rangle^{2}+\mathrm{C}_{1}
\end{aligned}
$$

- Take note here that $\left\langle x-\frac{\mathrm{L}}{2}\right\rangle$ is integrate as a function of $x$. This is rooted to advanced math that $M$ acaulay use in his method that need to be remember.
- From the displacement integration :

$$
\begin{gathered}
v E I=\iint \frac{\mathrm{P}}{2} x-\mathrm{P}\left\langle x-\frac{\mathrm{L}}{2}\right\rangle d x \\
v E I=\int \frac{\mathrm{P}}{4} x^{2}-\frac{\mathrm{P}}{2}\left\langle x-\frac{\mathrm{L}}{2}\right\rangle^{2}+\mathrm{C}_{1} d x \\
v E I=\frac{\mathrm{P}}{12} x^{3}-\frac{\mathrm{P}}{6}\left\langle x-\frac{\mathrm{L}}{2}\right\rangle^{3}+\mathrm{C}_{1} x+\mathrm{C}_{2}
\end{gathered}
$$

- Again Ttke note here that $\left\langle x-\frac{L}{2}\right\rangle$ is integrate as a function of $x$.
- From slope and displacement integration procedure, 2 unknown were obtained and solved using the boundary condition:
$-v=0, x=0$
$-v=0, x=\mathrm{L}$
- Please remember :

$$
\left\langle x-\frac{\mathrm{L}}{2}\right\rangle=\left\{\begin{array}{cc}
0, & x \leq \frac{\mathrm{L}}{2} \\
x-\frac{\mathrm{L}}{2}, & x>\frac{\mathrm{L}}{2}
\end{array}\right.
$$

- Lets use the first boundary

$$
-v=0, x=0
$$

$$
0 E I=\frac{\mathrm{P}}{12} 0^{3}-\frac{\mathrm{P}}{6}\left\langle 0-\frac{\mathrm{L}}{2}\right\rangle^{3}+\mathrm{C}_{1} 0+\mathrm{C}_{2}
$$

- Inside the bracket $\left\langle x-\frac{L}{2}\right\rangle=-\frac{\mathrm{L}}{2}=0$

$$
0 E I=\frac{\mathrm{P}}{12} 0^{3}-0+\mathrm{C}_{1} 0+\mathrm{C}_{2}
$$

$$
C_{2}=0
$$

- The second boundary
$-v=0, x=$ Land $\mathrm{C}_{2}=0$

$$
0 E I=\frac{\mathrm{P}}{12} \mathrm{~L}^{3}-\frac{\mathrm{P}}{6}\left\langle\mathrm{~L}-\frac{\mathrm{L}}{2}\right\rangle^{3}+\mathrm{C}_{1} \mathrm{~L}
$$

- Inside the bracket $\left\langle L-\frac{L}{2}\right\rangle=\frac{\mathrm{L}}{2}$ we use the value

$$
\mathrm{C}_{1}=-\frac{3 \mathrm{PL}^{2}}{48}
$$

- Using unknown:

$$
\begin{aligned}
-\mathrm{C}_{1}= & -\frac{3 \mathrm{PL}^{2}}{48} \text { and } \mathrm{C}_{2}=0 \\
& \theta E I=\frac{\mathrm{P}}{4} x^{2}-\frac{\mathrm{P}}{2}\left\langle x-\frac{\mathrm{L}}{2}\right)^{2}-\frac{3 \mathrm{PL}^{2}}{48} \\
& v E I=\frac{\mathrm{P}}{12} x^{3}-\frac{\mathrm{P}}{6}\left(x-\frac{\mathrm{L}}{2}\right)^{3}-\frac{3 \mathrm{PL}^{2}}{48} x
\end{aligned}
$$

- This equation can be use to obtain deflection and displacement at any position of the beam following 'turn off' rule.
- Lets determine slope at the support :
- At $x=0$

$$
\theta E I=\frac{\mathrm{P}}{4} 0^{2}-\frac{\mathrm{P}}{2}\left\langle 0-\frac{\mathrm{L}}{2}\right\rangle^{2}-\frac{3 \mathrm{PL}^{2}}{48}
$$

- Inside the bracket $\left\langle x-\frac{\mathrm{L}}{2}\right\rangle=-\frac{\mathrm{L}}{2}=0$

$$
\begin{aligned}
\theta E I & =-\frac{3 \mathrm{PL}^{2}}{48} \\
\theta & =-\frac{\mathrm{PL}^{2}}{16 E I}
\end{aligned}
$$



- Lets determine slope at the support :
- At $x=\mathrm{L}$

$$
\theta E I=\frac{\mathrm{P}}{4} \mathrm{~L}^{2}-\frac{\mathrm{P}}{2}\left\langle\mathrm{~L}-\frac{\mathrm{L}}{2}\right\rangle^{2}-\frac{3 \mathrm{PL}^{2}}{48}
$$

- Inside the bracket $\left\langle x-\frac{\mathrm{L}}{2}\right\rangle=\frac{\mathrm{L}}{2}$

$$
\begin{aligned}
& \theta E I=\frac{3 \mathrm{PL}^{2}}{48} \\
& \theta=+\frac{\mathrm{PL}^{2}}{16 E I}
\end{aligned}
$$



- Lets determine maximum displacement at the midspan :
- At $x=\frac{\mathrm{L}}{2}$

$$
v E I=\frac{\mathrm{P}}{12}\left(\frac{\mathrm{~L}}{2}\right)^{3}-\frac{\mathrm{P}}{6}\left\langle\frac{\mathrm{~L}}{2}-\frac{\mathrm{L}}{2}\right\rangle^{3}-\frac{3 \mathrm{PL}^{2}}{48}\left(\frac{\mathrm{~L}}{2}\right)
$$

- Inside the bracket $\left\langle x-\frac{L}{2}\right\rangle=0$



## M acauly's M ethod

In this example we take a beam with the UDL of $20 \mathrm{kN} / \mathrm{m}$ applied to the centre of the beam as shown. The beam has the materials property, $\mathrm{E}=30 \mathrm{kN} / \mathrm{mm}^{2}$ and a cross section in mm as shown. Determine the maximum displacement in the beam


$$
\begin{aligned}
& I=\frac{b d^{3}}{12}=\frac{200 \cdot 600^{3}}{12}=36 \times 10^{8} \mathrm{~mm}^{4} \\
& E I=\frac{(30)\left(36 \times 10^{8}\right)}{10^{6}}=108 \times 10^{3} \mathrm{kNm}^{2}
\end{aligned}
$$




- Taking moments about the cut, we have:

$$
M(x)-40 x+\frac{20}{2}[x-2]^{2}-\frac{20}{2}[x-6]^{2}=0
$$

- Again the M acaulay brackets (take note here $\langle\quad\rangle=[\quad]$ have been used to indicate when terms should become zero. Hence:

$$
M(x)=40 x-\frac{20}{2}[x-2]^{2}+\frac{20}{2}[x-6]^{2}
$$

- Applying Euler-Bernoulli ( $v=y$ ):

$$
M(x)=E I \frac{d^{2} y}{d x^{2}}=40 x-\frac{20}{2}[x-2]^{2}+\frac{20}{2}[x-6]^{2}
$$

Equation 1

- Integrate Equation 1 to get the slope equation

$$
E I \frac{d y}{d x}=\frac{40}{2} x^{2}-\frac{20}{6}[x-2]^{3}+\frac{20}{6}[x-6]^{3}+C_{\theta}
$$

Equation 2

- Integrate Equation 2 to get the displacement equation

$$
E I y=\frac{40}{6} x^{3}-\frac{20}{24}[x-2]^{4}+\frac{20}{24}[x-6]^{4}+C_{\theta} x+C_{\delta}
$$

Equation 3

- The boundary conditions are:
- Support A: $y=0$ at $x=0$
- Support B: $y=0$ at $x=8$
- So for the first boundary condition:

$$
\begin{array}{r}
E I(0)=\frac{40}{6}(0)^{3}-\frac{2 \theta}{24}[\theta-2]^{4}+\frac{2 \theta}{24}(\theta<-6]^{4}+C_{\theta}(0)+C_{\delta} \\
C_{\delta}=0
\end{array}
$$

- For the second boundary condition:

$$
\begin{aligned}
E I(0) & =\frac{40}{6}(8)^{3}-\frac{20}{24}(6)^{4}+\frac{20}{24}(2)^{4}+8 C_{\theta} \\
C_{\theta} & =-293.33
\end{aligned}
$$

- Insert constants into Equations 3
(displacement)

$$
E I y=\frac{40}{6} x^{3}-\frac{20}{24}[x-2]^{4}+\frac{20}{24}[x-6]^{4}-293.33 x
$$

$$
\begin{aligned}
E I \delta_{\max } & =\frac{40}{6}(4)^{3}-\frac{20}{24}(2)^{4}+\frac{2 \theta}{24}[-6]^{4}-293.33(4) \\
& =-760 \\
\delta_{\max } & =\frac{-760}{E I}=\frac{-760}{108 \times 20^{3}}=-0.00704 \mathrm{~m} \\
\delta_{\max } & =-7.04 \mathrm{~mm}
\end{aligned}
$$

- This is therefore a downward deflection as expected.



## THANKS

by Saffuan Wan Ahmad

## Author Information

Mohd Arif Bin Sulaiman<br>Mohd Faizal Bin Md. Jaafar<br>Mohammad Amirulkhairi Bin Zubir<br>Rokiah Binti Othman<br>Norhaiza Binti Ghazali<br>Shariza Binti Mat Aris

